Dynamic Modeling and Controlling Ammonia Fueled SOFCC-GT Hybrid System for Commercial Aviation

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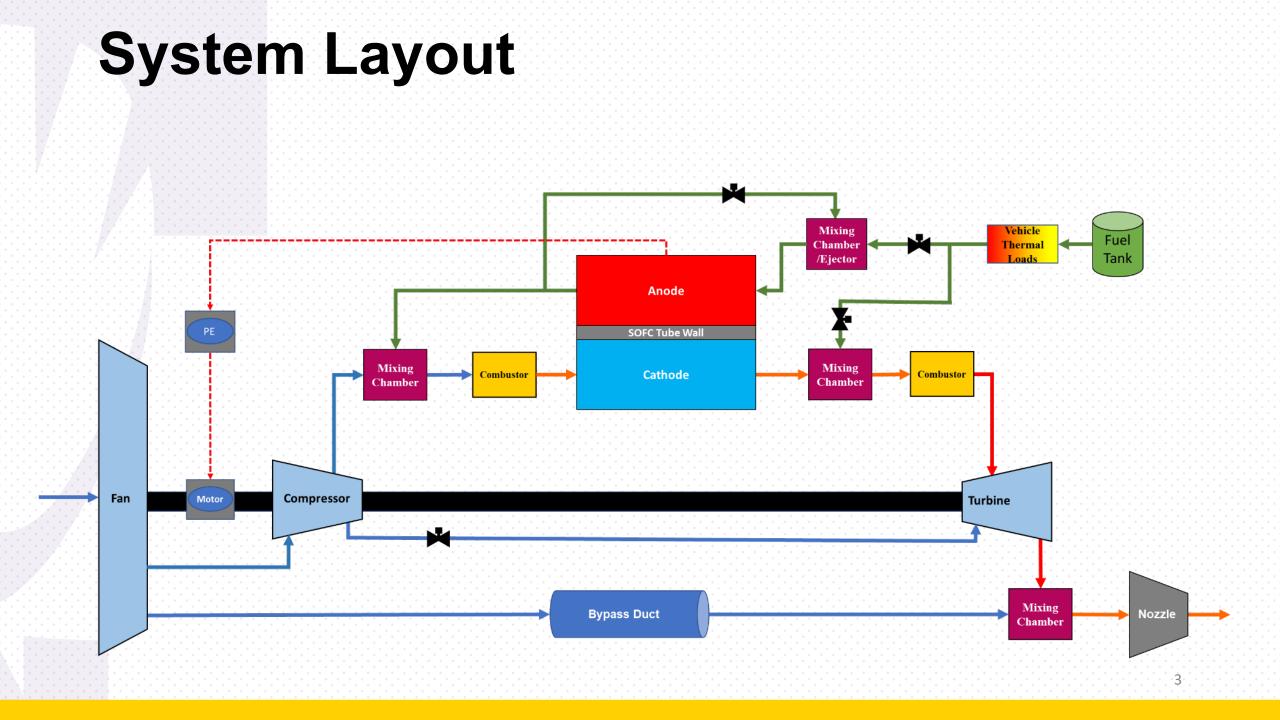




Background

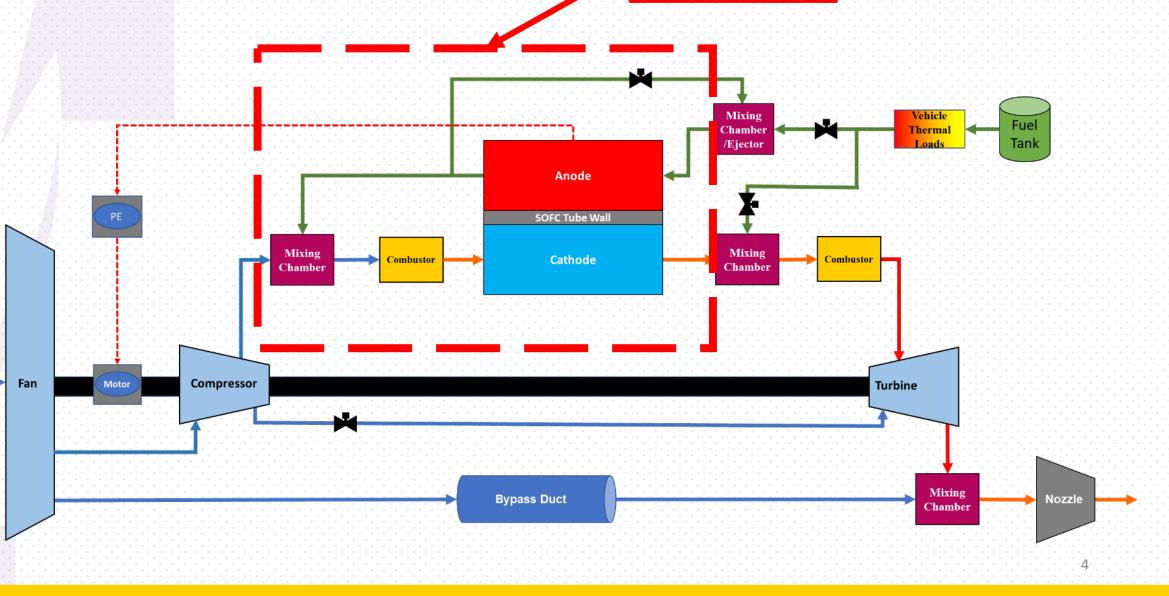
- Develop >150 Passenger Commercial Aircraft with Zero Carbon Emissions and Minimal NOx Emissions
- Ammonia(*NH*₃) Used as Fuel to Eliminate Carbon Emissions
- Intended to Retrofit Boeing 737
 Aircraft for Hybridization
- Solid Oxide Fuel Cells (SOFC's) used as Alternative to Batteries for Electric Power

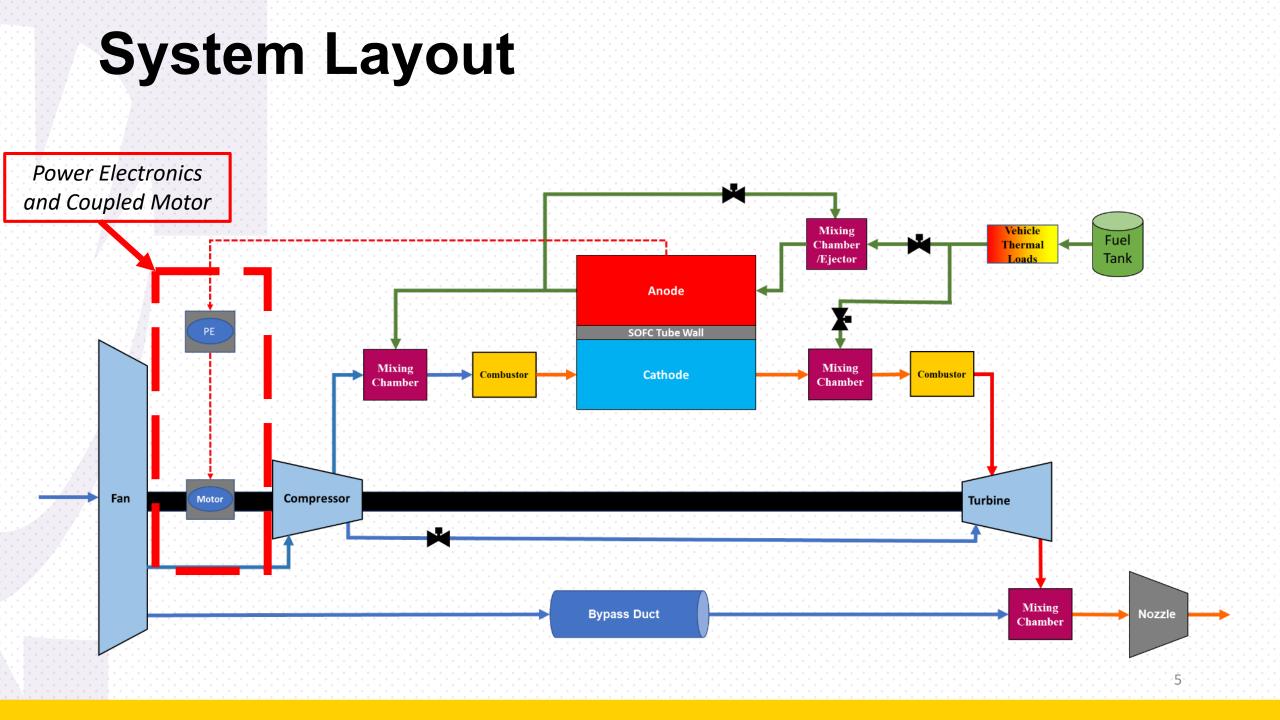




System Layout

Solid Oxide Fuel Cell Combustor





System Overview

Advantages

- Increased System Efficiency
 Increased System Mass
- Zero Carbon Emissions
- Lower NOx due to Lower Turbine Inlet Temperature
- Similar to Legacy Equipment

Increased Complexity

Disadvantages

New Environmental Concerns

6

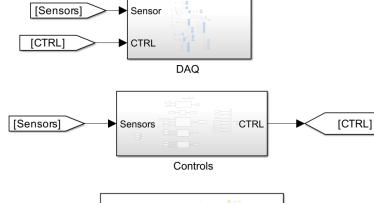
Fuel Storage Challenges

System Modeling

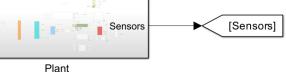
- Dynamic Model created using SIMULINK environment
- Conservation Equations Applied
 - Mass
 - Energy
 - Species

Per Wing Basis

- Controls for warm-up and operation
- Control Loop is to varies TIT to match given Thrust Profile

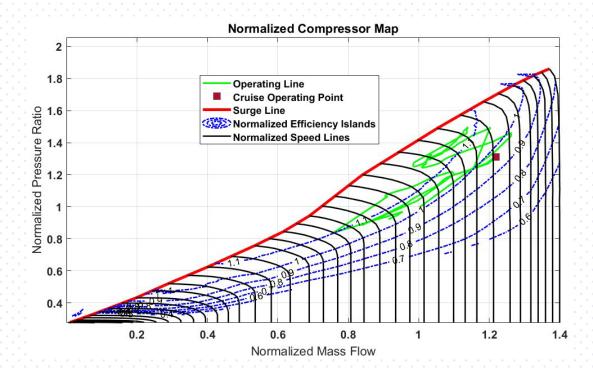


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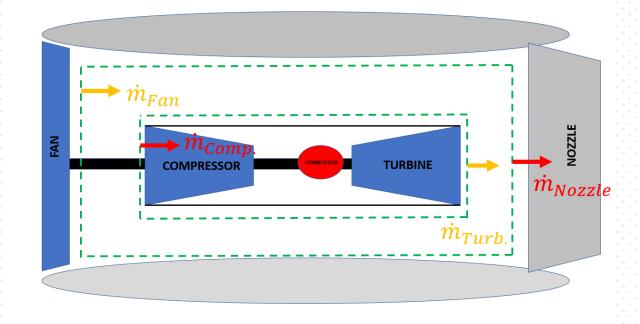
Turbomachinery

- Normalized Maps used for Turbomachinery
- Fan and Compressor Utilize
 Identical Performance Map
- Isentropic Relations used to Determine Power based on Pressure Ratio



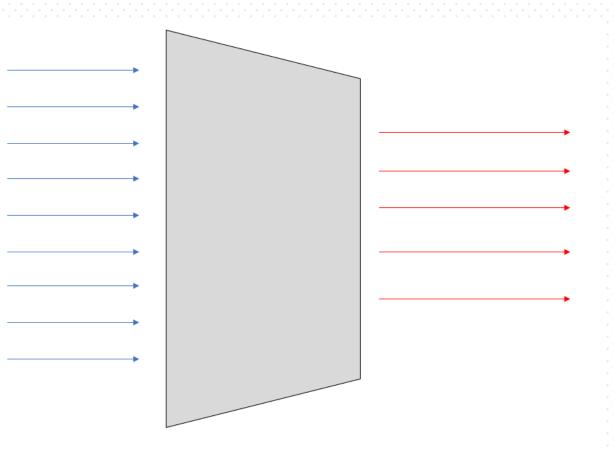
Bypass Duct

- Bypass Duct treated as a
 Plenum Volume
- Static Pressure Determined
 using Ideal Gas Law
- Dynamic Pressure found
 Using Assumed Flow Area



Nozzle

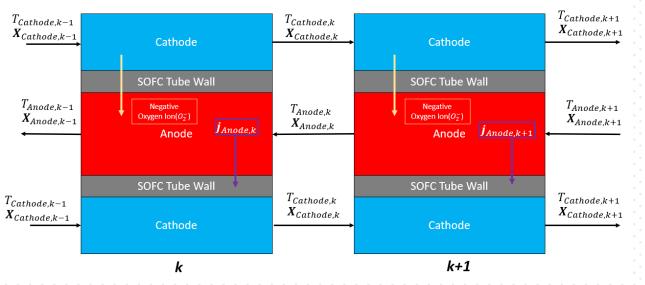
- Converging Nozzle
 Area Design for Mach <= 0.9 at Exit for Noise
- Choked flow condition continuously checked to determine if flow is Choked at Nozzle Exit



SOFC Tubes

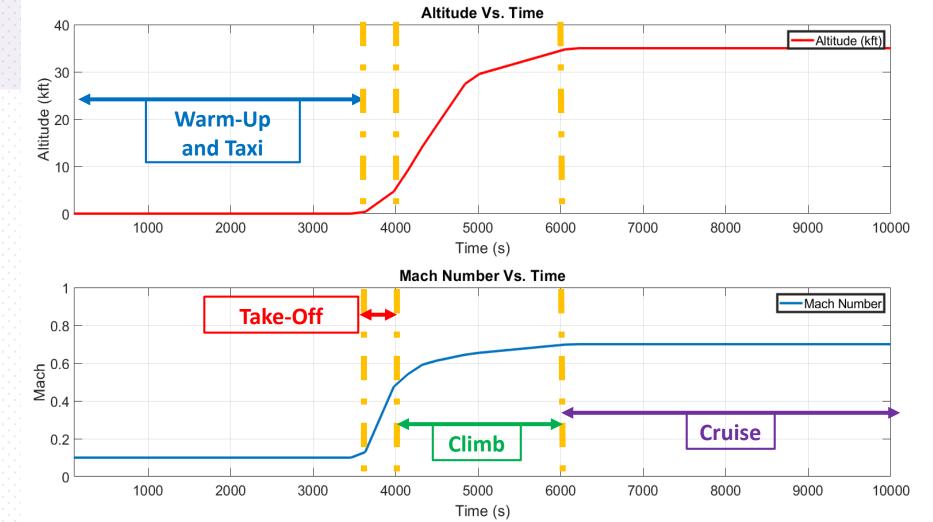
- Tubular SOFC's are discretized into Nodes
- 1-D Conservation Equations
 applied at each Node
 - Temperature
 - Species
 - Current

 Ammonia Cracked before Entering SOFC to Minimize Thermal Gradients

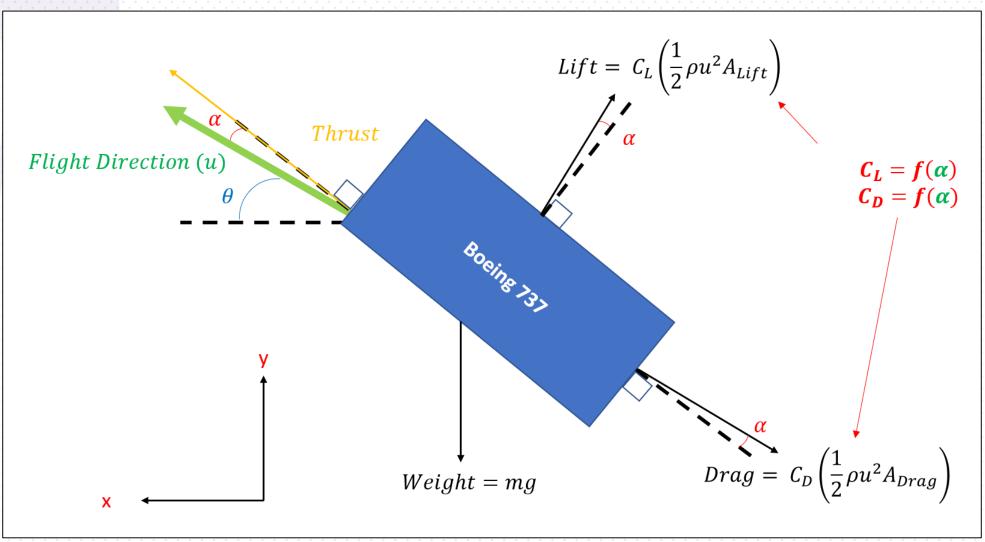




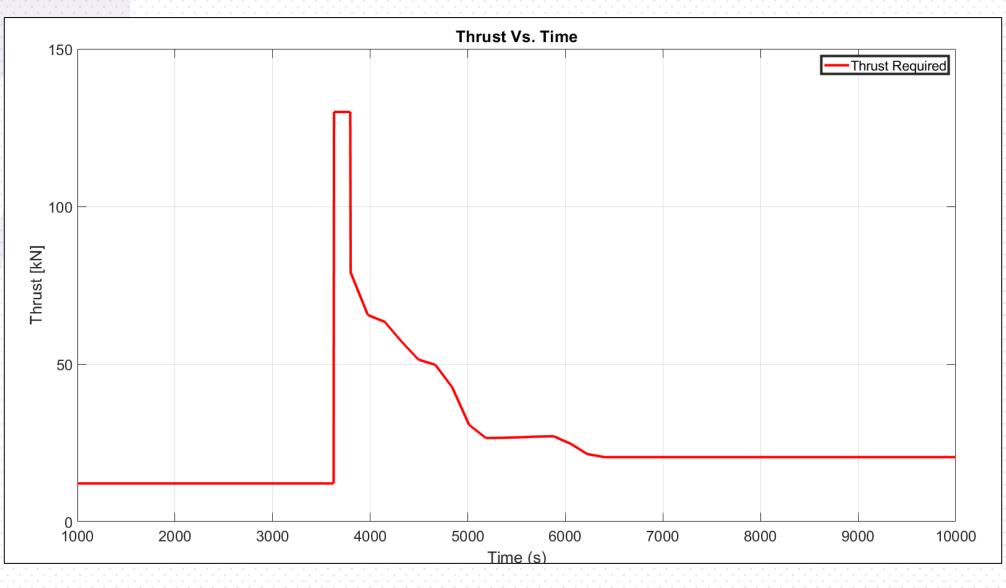
Flight Profile



2DOF Formulation

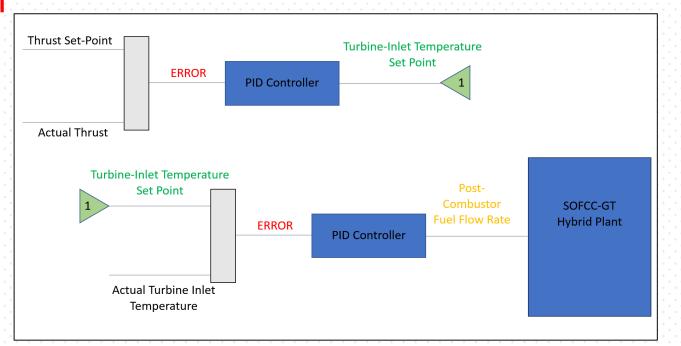


2DOF Formulation

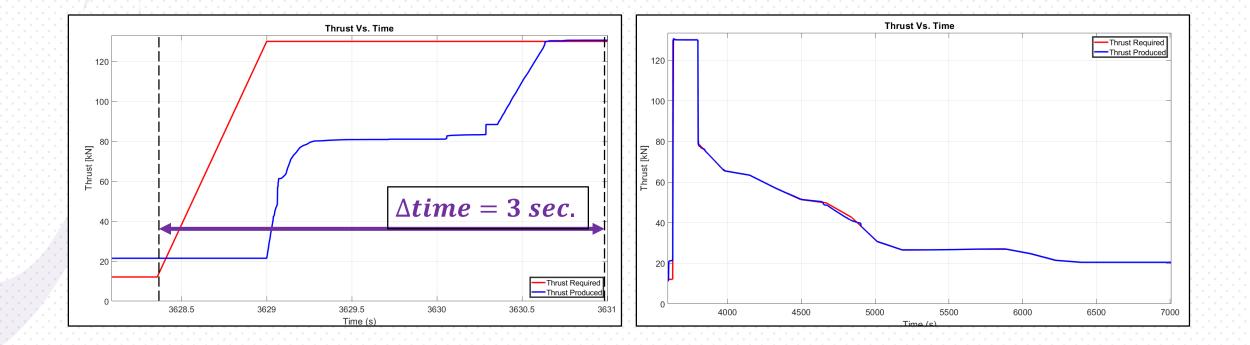


Operational Control Strategy

- SOFC Operated at Maximum Power throughout Flight Profile
- Turbine Inlet Temperature is adjusted via Fuel Flow Rate to Match System Thrust to 2DOF Predicted Thrust

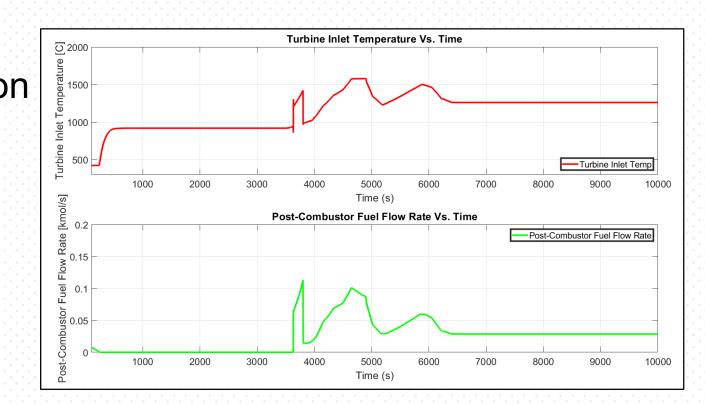


Results (Thrust)



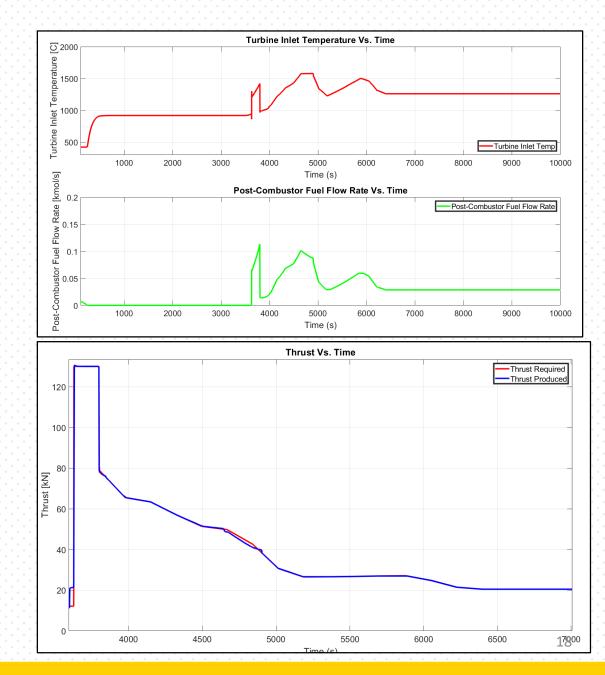
Results (Turbine Inlet Temperature)

- Turbine Inlet Temperature Reduced Operates at ~1,260°C at Cruise Condition
- Fuel flow calculated across flight profile with cruise fuel flow being ~0.03 kmol/s



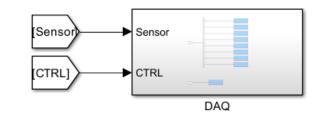
Conclusions

- Hybrid System Using Ammonia Fed SOFC Proposed as a Carbonless Propulsion System
- System was able to Dynamically react and control to match Thrust requirement along flight profile
- Turbine Inlet Temperature was able to be reduced during cruise and take-off, reducing NOx formation



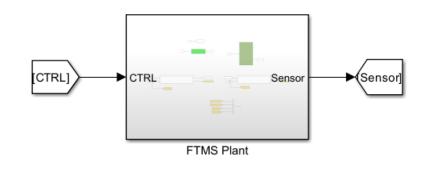
Future Work

- 2DOF incorporation into controls of system model
- Implement Fuel Thermal Management System (FTMS) into system model
- Investigate addition of batteries onboard for high load transients
- Optimize system sizing and controls to maximize range









Acknowledgements

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Thank you!

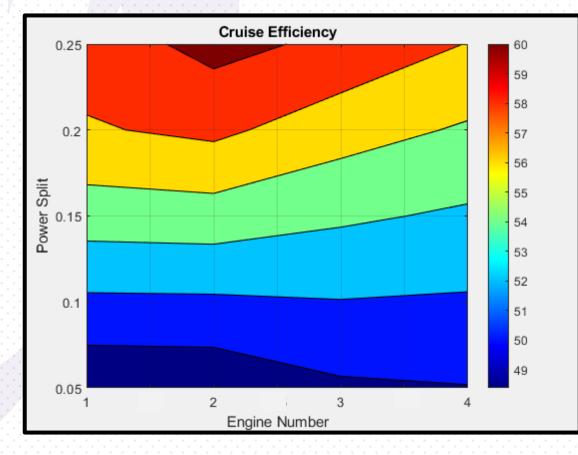
Are there any questions?

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CarbonLess Electric AviatioN

Efficiency



₩_{SOFC} $Power_{Split} = \frac{1}{\dot{W}_{SOFC} + \dot{W}_{Turbine} - \dot{W}_{Compressor}}$

Cruise Propulsive Efficiency

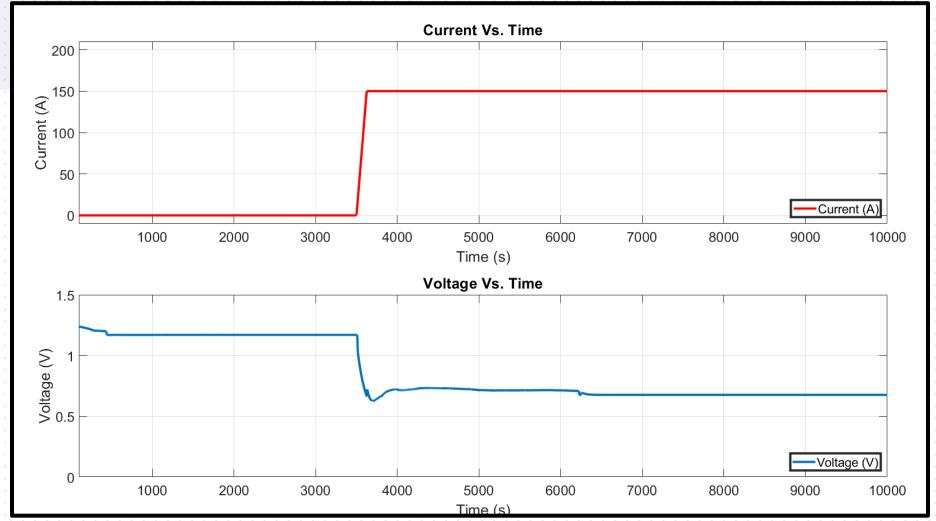
 Leap 1B
 SOFCC-GT

 ~36%
 >52%

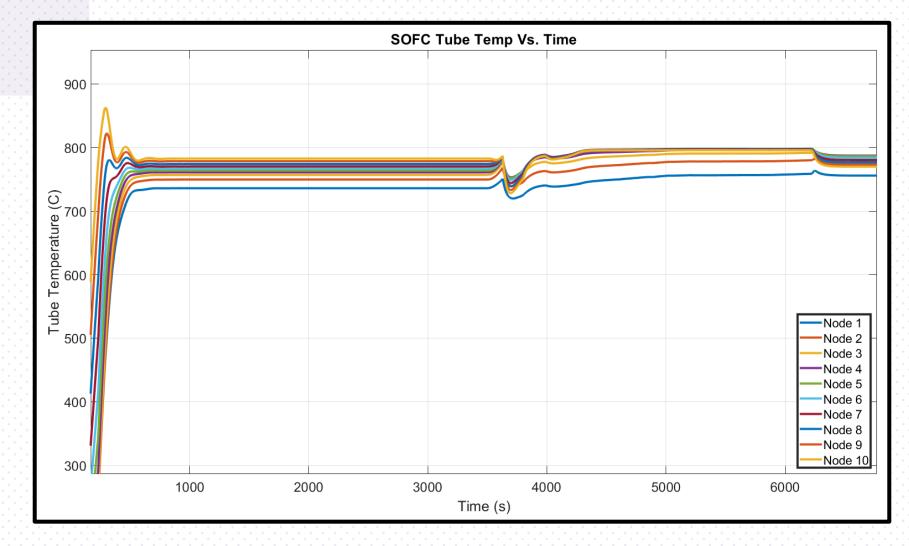
SOFCC ≈ 1.4X

Efficient

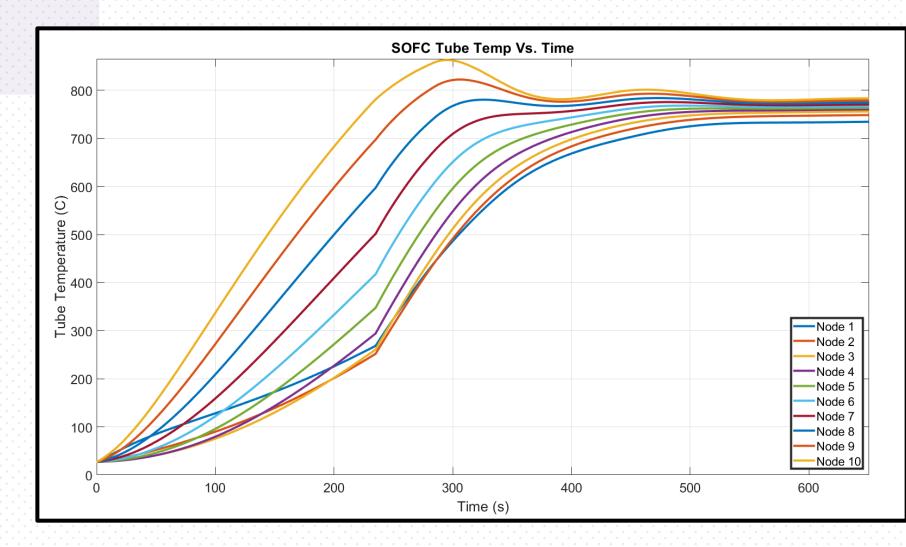
Voltage and Current



Tube Temps



Tube Temps (Warm-Up)



Power

